# The Rise and Fall of CBM

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#### **Plan of Presentation**

#### → Expectations

- → Technological progress (sensors, techniques, methods etc.)
- Barriers to implementation
- → CBM/PHM
- → Risk assessment
- → Modelling failure
- → Taxonomy
- → Sensor fusion
- Jet Engine lubrication system example
- Concluding remarks



### Background

- → Defence equipment and systems function in harsh environmental and operational conditions and must meet stringent requirements of reliability, safety, availability and maintainability – particularly with the introduction of performance based contracts (PBC)
- → To reduce the high cost of development for new products, OEMs use a vast array of computer aided techniques during the design and testing stages
- Maintainability requirements, long ignored by designers and OEMs, has assumed great importance and forced a rethinking of the way the design of new systems should be carried out
- → Availability is a major constraint, and it has became important to develop techniques to monitor the health of a system, to diagnose system problems prior to its failure and to prognose the system's remaining life
- → Efforts have been made to justify these new design approaches with a business case that reflects PBC requirements and the current US DoD system acquisition policies that focus on the cost of sustainment over the entire system life cycle
- Maintenance Technology has become recognized as an academic discipline





#### **Expectations**

- Enhanced system reliability and equipment safety
- Reduced maintenance manpower, spares and repair costs
- → Eliminate or significantly reduce scheduled inspections
- → Maximised lead time for maintenance and parts
- → Procurement
- → Automatically isolated faults
- → Real time notification of an upcoming maintenance event for the logistics chain
- > Catch potentially catastrophic failures before they occur
- Detect and monitor incipient faults until just prior to failure



### The Rise of CBM

- → Rapid progress was made in the 1960s 1980s in the development of new sensors, symptom monitoring techniques and performance monitoring in aircraft, marine, railways and mining machinery applications
- During this period, monitoring techniques were seldom used together to provide comprehensive and reliable detection and diagnosis of failures
- Likewise, research on detection and diagnostic techniques and methodology was usually directed towards a single technique; for example: vibration monitoring
- → The situation changed in the early eighties when the concept of On-Condition Maintenance was developed [over the years the name changed to Condition Based Maintenance (CBM)] and applied in high risk industries like aviation, mining and offshore oil production
- → Since the 1990s there has been significant progress in the development of new sensing techniques, diagnostic and prognostic methodologies and the application of computer analysis techniques



#### **Technical Barriers to effective CBM**

The Advanced Technology Program (ATP), of the National Institute of Standards and Technology (NIST), held a workshop on Condition-Based Maintenance (CBM) as part of it's November 17-18, 1998 Fall Meeting in Atlanta.

Discussions with companies identified 3 technical barriers to CBM's widespread implementation:

- The inability to accurately and reliably predict the remaining useful life of a machine (*prognostics*)
- → The inability to continually monitor a machine (sensing)
- → The inability of maintenance systems to learn and identify impending failures and recommend what action should be taken (*reasoning*).

These barriers could potentially be addressed through innovations in three technical areas:

- Prognostication capabilities
- Cost effective sensor and monitoring systems
- → Reasoning or expert systems



## **Systemic Barriers to effective CBM**

- → In many military/industrial applications the metrics for evaluating successful implementation of CBM are not clearly defined (*risk, economics, performance*)
- → Lack of clear guidelines or business case for when and why CBM is preferred to other maintenance approaches (technical/economic)
- → CBM programs are initiated without full knowledge of how the system can fail
- → The effectiveness of a CBM program cannot be evaluated with current management tools.
- Maintenance requirements/specifications are not defined at the concept formulation stage of the design process
- → Identification of an optimum level of diagnostic and prognostic requirements and specifications is not generated
- Selection of an optimum monitoring mix (selection of sensors) should be system oriented but is often driven by the vendors of sensors
- → Maintenance management systems are inadequate
- → Historical data, postmortem results not available or accessible
- Uncertainty of ROI (Plant Services Magazine (USA)"...In a survey of 500 companies, less than 3% of respondents were able to achieve a measurable return on their investment in Predictive Maintenance technologies")



### **Knowledge Barriers to effective CBM**

- Limited or no knowledge retention about CBM within the OEM or customer
- → Skills issues are not addressed
- → Education of CBM managers/engineers not available at Universities (Monash University had undergraduate/postgraduate programs in CBM supported by multidisciplinary laboratory from 1980 to 1996)
- Widespread research in CBM but it is invariably directed towards specific techniques (better mousetrap symptom)

Google "Condition based Maintenance" - 33,000,000 hits! Google "Condition based Maintenance barriers" - 1,080,000 hits!. Google "Gearbox Condition monitoring" - 44,000 hits Google "Bearings Condition monitoring" - 350,000 hits Google "Vibration Condition monitoring" - 351,000 hits Google "Contamination Condition monitoring" - 304,000 hits



# CBM / PHM / RCM and other TLAs

- The methodologies and approaches of CBM evolved from On-Condition Maintenance (OCM)
- Reliability Centred Maintenance (RCM) is a variant of the CBM approach
- → PHM is a further evolution of the CBM concept, and is also sometimes referred to as Vehicle Health Management (VHM)







## The PHM cycle

An effective PHM implementation for a system requires two main cycles of development: design and operation

- The Design Cycle is required in order to generate the knowledge base from which the PHM system can obtain its decisions.
- The Operation Cycle describes the steps taken within the PHM system from detection of faults through to conveying instructions or actions.





#### **CBM/PHM** - what are we dealing with?





#### Interaction of MAD and CBM/PHM Layers at Design Stage





# **Case Study - Mining**

- → 12 mine sites mining trucks, conveyors, shovels etc.
- → Data from mines' maintenance management systems
- → Approx 500 MB of data collected over period of up to 5 years
- → Limited number of detection/diagnostic/ techniques
- → External contractors no in-house knowledge
- → Only 4 sites had useful information although incomplete

#### Conclusions (from the report):

- → Sampling/detection and diagnosis do not follow the **best practice** to achieve meaningful indication of machine state
- → Any reporting should have have **deliverables**, or information will not be useful.
- → Unspecified conditions before failure occurred
- → Lack of information of how the system, component, part failed ie. postmortem
- → Outline the reactive and pro-active activities.
- Unknown or missing grade and quality of roads, drivers, trained, gender, mechanics, conditions, weather, material being hauled, oil used, petrol used, original parts used, shift work, 7 day week, support, underground, humidity, walk around each day, same route etc.
- → Effective FMEA/FMECA Analysis should be conducted prior to monitoring
- → No visible CBM design/plan
- → No possibility to assess ROI





# Subsea / Aerospace

#### Risks

- → Severe operating environment
- → Stringent statutory safety standards
- → Safety critical systems
- → Expensive maintenance
- → Long innovation lead time
- → High technology
- → Conservative attitudes
- → High reliability requirements
- → Single shot operations
- → Very high cost of failure

#### Tools to deal with risks

- > Computer based design methods
- Reliability and Hazard Analysis
- → Failure analysis (FMECA/FTA)
- PHM (Prognostics and Health Management)

**PHM**Technology

- → Condition Monitoring CBM
- Testing





### **Risk reduction – CBM/PHM**



#### What is it?

- → Risk assessment using techniques like FMECA, HAZOP, RCM etc.
- Diagnostics is the process of determining the state of a component to perform its function(s)
- → Prognostics is predictive diagnostics which includes determining the remaining life or time span of proper operation of a component
- Health Management is the capability to make appropriate decisions about maintenance actions based on diagnostics/prognostics information, available resources and operational demand.



#### **Criteria for RCM Processes**

# SAE JA1011 "Evaluation Criteria for RCM Processes" defines seven questions for RCM:

- What are the functions...of the asset...(functions)?
- In what ways can it fail...(functional failures)?
- What causes each functional failure (failure modes)?
- What happens when each failure occurs (failure effects)?
- In what way does each failure matter (failure consequences)?
- What should be done...(proactive tasks and intervals)?
- What should be done if a suitable proactive task cannot be found?



#### **Risk Assessment - FMECA**





#### **Risk Assessment – e.g. FMECA**

#### Why FMECA is carried out

- → Statutory requirement must be done
- → We need to have audit trail in case of problems
- A need to know of how to improve system safety
- → The integrator insisted on it
- → Reliability people need it

#### Why FMECA should be carried out

- We need to know what to monitor and what sensors to use
- We need to have capability to detect, diagnose and prognose the state of the system
- → To design-out failures
  - We need to know how the system can fail so we are prepared to deal with it
- To enhance diagnostic capabilities





### **Reasons for failure of Risk Assessment**

- Dependencies of failures not identified spreadsheet vs model based
- Inadequate Identification of Risks functional failures (failure modes) vs physical failures
- Incomplete database of failures (deficient FMECA)
- Taxonomy confusion what is the cause, mechanism of failure, fault, symptom and/or failure mode
- Symptom vs Syndrome approach
- → Sensor fusion not based on failures dependencies (fall-back testability)
- Diagnostic rules not based on dependencies
- → Reliability of Hardware not the same as Functional Reliability
- > Different models for Criticality and Reliability Assessment





#### **Risk reduction or is it?**



- Risk is still there if failures are missed
- We cannot design a diagnostic system without knowledge of failures
- We do not really know what we should monitor
- Sensors cover only identified failures



#### **Dependencies Modelling**





### Fault propagation - dependability

- $\rightarrow$  All faults are enumerated.
- → Transient and steady-state responses to faults are identified





#### **Model of a Pump**





#### **Modelling of failure**



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4-6 October 2010

#### **Taxonomy problems**

Source - Item	Failure term	Cause	Mechanism	Fault	L/S	FF	No Class	LIF	
RAC - failure mod	es								
Pump, hydraulic	Leaking			X					
	Improper flow					X			
	No flow					X			L/S=Loss/Sympton
Electric Motor, AC	Winding failure					X		X	
	bearing failure					X		х	FF=Functional Failure
	fails to run, after start					X			
	fails to start					X			LIF=Lower Indenture Level Failure
NSWC - failure mo	odes								
Electric Motor, AC	worn bearing			X				X	
	open winding			X				X	
	shorted winding			Х				X	
	cracked housing			Х				X	
	sheared armature shaft			X				X	
	cracked rotor laminations			Х				X	
	worn brushes			X				X	
	worn sleeve bearing			X				X	
Pump, hydraulic	Pump cavitation		X						
	component corrosion		X						
	Low net postive suction head	X							
	shaft unbalance			X				X	
	external leakage			X				X	
	mechanical noise				X				
	positive suction head to low	X							
	pump discharge head to high					X			
	suction line clogged	X							
	pressure surges	X							
	increased fluid temperature	X							





#### **Taxonomy problems**

Eile E	IQ-RM PRO - APIS IT GmbH [10000-50] - Personal Desktop										
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	FMEA Forms Ed	Failure Mo- des	Effects	42 - si S	Gnal cat	Causes	Sign [Design] ) Preventive Ac- tion	0	Detection Ac- tion	D	RPN
■	Solderability of shielding	E inadequate solderability of shielding	does not resist environmental conditions over life-time			Yerong shielding material chosen	experience from earlier development projects	3	material test with plate sample	5	R ^
	sensitivity against interferences of signal transmission (both level and modulation)	does not proctect the conductor from interfering signals from environment	Signal does not represent the input values correctly			Shielding material chosen	experience from earlier development projects	3	Q material test with plate sample	5	



### **Identifying Risk**





#### Failures - Symptoms/Syndromes





#### **Sensor selection**





#### **MADe software**



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**PHM**Technology

#### **MADe software**





#### **RR250 Engine Lubrication System**





#### **Engine Lubrication System Model**





#### **Define Component Structure**





#### **Define Component Functions**

📄 Oil Gear Pressure Pump 🔯

#### **Functions**

Functions

Functions

Flows

#### Function Definition

Drag Functions from the Functions Taxonomy (Functions Tab - switch using the tabs at the top of the Functions tree) into the empty space of the box below. Then using the Flow taxonomy (Flows tab) drag an appropriate Flow from the Flows Tree into a Function.





#### **Define Physical Failures**





#### **Propagate Functional Failures >> Dependency**



<sup>4-6</sup> October 2010

#### **Assess Criticality**

#### 📩 Lubrication system for RR250-20C Engine 🔀 🛛

∓ End Effect Item: 📑 Oil Nozzle B7





#### **Produce FMEA/FMECA Report**

Oil Gear Press	ure Pump	🗈 Lubrication	system for RR	250-20C Engine	e Report 🔀										- 5
	Input Shaft		N/A	Vibration loading	High cycle fatigue	Fractured	Very low (Input Shaft_Coupling)	Low (Oil Nozzle B7)		10.0	10.0	2.3	230		^
ITEM INDENTUR NEXT INDENTUR SYSTEM NAME REFERENCE DR/	RE LEVEL RE LEVEL AWING	1 Drive Flex Sh Lubricat	Shaft aft Coupling tion system for RR	250-20C Engine				LOADING TY LOAD CHAR LOADING TII	PES ACTERISTIC ME	Torsion Constant Low - Le	: load ss or equal to miss	sion time.			
Item ID	Item Name	Operation Mode	Function	Cause	Failure		Effects		Detection		Critic	cality _		Remarks	
	1 Drive Shaft		N/A	Cyclic mechanical loads	High cycle fatigue	Fractured	Very low (Coupling_1 Drive Shaft)	Low (Oil Nozzle B7)	Method	10.0	s 10.0	2.3	230		
	1 Drive Shaft		N/A	Vibration loading	High cycle fatigue	Fractured	Very low (Coupling_1 Drive Shaft)	Low (Oil Nozzle B7)		10.0	10.0	2.3	230		]
ITEM INDENTUR NEXT INDENTUR SYSTEM NAME REFERENCE DR	RE LEVEL RE LEVEL AWING	Couplin Flex Sh Lubricet	9 aft Coupling tion system for RR	250-20C Engine				LOADING TY LOAD CHAR/ LOADING TIP	PES ACTERISTIC ME	None Constant High - M	: load ultiple missions.				_
Item ID	Item Name	Operation Mode	Function	Cause	Failure Mechanism	Local	Effects	End	Detection Method	0	Critic	cality D	RPN	Remarks	
	Coupling		N/A	Vibration loading	High cycle fatigue	Fractured	Very low (Input Shaft_Coupling)	Low (Oil Nozzle B7)		1.0	10.0	2.3	23		1
	Coupling		N/A	Cyclic mechanical Ioads	High cycle fatigue	Fractured	Very low (Input Shaft_Coupling)	Low (Oil Nozzle B7)		5.0	10.0	2.3	115		
	Coupling		N/A	Random or transient mechanical loads	Compression fracture	Fractured	Very low (Input Shaft_Coupling)	Low (Oil Nozzle B7)		7.5	10.0	2.3	173		
	Coupling		N/A	High mechanical load	Compression fracture	Fractured	Very low (Input Shaft_Coupling)	Low (Oil Nozzle B7)		10.0	10.0	2.3	230		
	Coupling		N/A	Vibration loading	High cycle fatigue	Fractured	Very low (Coupling_1 Drive Shaft)	Low (Oil Nozzle B7)		1.0	10.0	2.3	23		
	Coupling		N/A	Cyclic mechanical Ioads	High cycle fatigue	Fractured	Very low (Coupling_1 Drive Shaft)	Low (Oil Nozzle B7)		5.0	10.0	2.3	115		
	Coupling		N/A	Random or transient mechanical loads	Compression fracture	Fractured	Very low (Coupling_1 Drive Shaft)	Low (Oil Nozzle B7)		7.5	10.0	2.3	173		



#### **Assess hardware Reliability**





#### **Define Sensors Locations**

#### | 鼎 @ 🗋 💿 !: 📲 ! 🔍 • ! 📄 💿 📮 🏢 📳 !: 🖄 🌣 🖆 !! 🖬 📌 🀴 🏝

🛃 Lubrication system for RR250-20C Engine 🕅

#### ∓ End Effect Item: 💕 Oil Nozzle B7





#### Select sensors and generate diagnostic rules

Ws Help		
🚡 Lubrication system for RR250-20C Engine 🛛 🕞 sassas 🕱 🔪		- 6
		Interior in
Type: Dynamic Cost: \$0.00		Failure (Testability):
Coverage: 100% Weight: 0.0 Possible Coverage: 100% Reliability: 0.0	Reanalyze	Failure (Initial):
		Failure Propagation:
		Disabled Sensors:
Sensor Allocations		Sensor Library 😗 Edit
Locations Flows		To allocate Sensors drag a required Sensor from the Library on to the appropriate location in the Sensor Allocations Area.
Flex Shaft Coupling (1)		Search
Sensors (1)		Ecce Sensor
Elow Properties (1)		🛓 🗀 Load Cells (Strain Gauge)
Mechanical - rotational Torque (1)		🖬 🗀 Load Cells (Hydraulic)
Symptoms (0)		Load Cells (Compensated)
Change in behaviour - 1 Drive Shaft (0)		B D Piezoelectric
Airborne noice - Counling 1 Drive Shaft (0)	<b>_</b>	Tactile Sensors
☑ Allow addition of a non-required sensor □ Show all hierarchy		✓ Filter based on element selection
Overview	^/	<u>N</u>
Ambiguity Groups III *Sensor Sets - Lubrication system for RP250-20	C Engine P Diagnostic Sets - sassas	
1219 Diagnostic Sets.		
Component	Rule	A
Oil Filter Bynass Check Valve	Ruic	
Oil Nozzles B5-6		
🗉 📦 Line Volume		
🖃 💼 Oil Gear Pressure Pump		
<ul> <li>Volumetric flow Pressure differential Decrease (TR)</li> </ul>		
	IF Sensor[Pump Connecting Volume] Pre	essure is Low AND Sensor[Flex Shaft Coupling] Torque is High
	THEN failure mode is Oil Gear Pressure F	Pump Volumetric flow Pressure differential Decrease (TR)
<ul> <li>Volumetric flow Pressure differential Decrease (SS)</li> </ul>		
	IF Sensor[Oil Filter] Volumetric flow is N	ominal AND Sensor[Flex Shaft Coupling] Torque is High AND Sensor[Check Valve] Volumetric flow is Nominal
	OR IF Sensor[Oil Filter] Volumetric flow	is Nominal AND Sensor[Check valve B7-8] Volumetric flow is Nominal
	OR IF Sensor[Pump Connecting Volume]	Pressure is Nominal AND Sensor[Check valve B7-8] Volumetric flow is Nominal
	OR IF Sensor[Oil Filter] Volumetric flow	is Nominal AND Sensor[Gearbox Lubrication arrangement] Volumetric flow is Nominal
	OR IF Sensor[Pump Connecting Volume]	Pressure is Nominal AND Sensor[Gearbox Lubrication arrangement] Volumetric flow is Nominal
	OR IF Sensor[Oil Filter] Volumetric flow	is Nominal AND Sensor[Line Volume B1] Pressure is Nominal
	OR IF Sensor[Pump Connecting Volume]	Pressure is Nominal AND Sensor[Line Volume B1] Pressure is Nominal
	OR IF Sensor[Oil Filter] Volumetric flow	Is Nominal AND Sensor[Line Volume B2-6] Pressure is Nominal
	OR IF Sensor[Pump Connecting Volume]	Pressure is Nominal AND Sensor[Line Volume B2-6] Pressure is Nominal
	OK IF Sensor[OII Filter] Volumetric flow	is Nominal AND Sensor[Manifold] Pressure is Nominal AND Sensor[Hex Shaft Coupling] Lorque is High
<		



#### **PHM Design Cycle Deliverables**

At the end of the risk assessment process, the user has knowledge of:

- → How the system can fail (failure modes)
- → How critical each failure is
- → What are the causes of functional failures
- What are the interactions between functional failures
- → What physical failures are linked to functional failure
- → Where to place sensors i.e sensor fusing
- How to monitor physical failures
- → How to diagnose functional failure
- → What is the expected reliability of the sensing system
- What is the expected functional and hardware reliability of the system





# **Concluding Remarks**

Despite expectations the acceptance and effectiveness CBM is in question. To be effective:

- → CBM/PHM programs must be designed and executed with the knowledge of the risks to which a system is exposed, i.e. the knowledge of <u>how the system fails</u>
- Model-based failure analysis, defining failures dependencies and improving the completeness of risk identifications, should be adopted in preference to checklists and "spreadsheet" based FMECA methodology or tools
- Model-based failure analysis should be adopted to enhance knowledge retention, knowledge transfer and to facilitate integration of risk assessment through supply chains
- Standardised taxonomies of functions, failure concepts and components should be adopted to improve readability/portability of risk assessment results
- Diagnostic rules and Sensors sets should be selected on the basis of the identified dependencies between failure modes (symptoms >> syndrome)
- Clear hierarchy of failure concepts should be enforced in the risk assessment process (cause > failure mechanism > fault > failure mode)
- → Physical failures (cause/failure mechanism/fault) and their symptoms should form the basis for BIT design

